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Ring Laser Gyro Strapdown Inertial Navigation System

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Ring Laser Gyro Strapdown Inertial Navigation System Introduction

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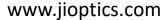
D70A series of inertial navigation systems can be widely used in tanks, all kinds of armored vehicles, self-propelled artillery, rockets, all kinds of unmanned aerial vehicles, aircraft, ships, automobiles, high-speed rail, unmanned vehicles, mobile satellite communication systems, drilling and other carrying platforms. Flight control, attitude stabilization, weapon stabilization platform, positioning and positioning, etc.

Technical features

Medium-level ring laser gyro and quartz accelerometer
Optional static or moving base self-alignment
Error parameters calibration and compensation in full temperature range
Optional diverse input interfaces for GNSS/Odometer/DVL
Configurable navigation modes

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Excellent environmental suitability Military standards

Typical application scenarios

Sea vehicle navigation
Under-water vehicle navigation and positioning
Positioning and north-finding for land vehicle
Stabilization and control for moving carrier
Attitude measurement for demanding applications

Main functions of the system

It has the function of outputting information such as carrier position, heading, attitude angle, angular rate and speed in real time;

It has working modes such as pure inertial navigation and INS/GNSS (including Beidou) integrated navigation;

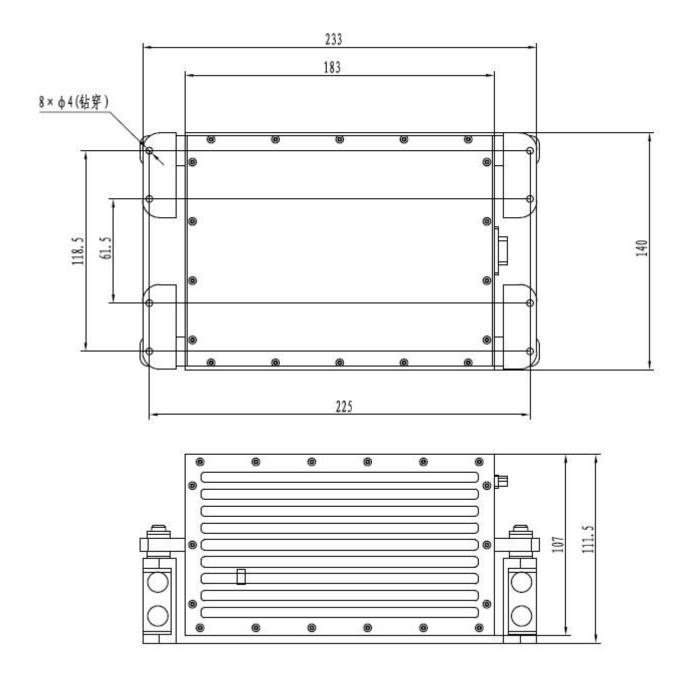
Possess the function of receiving satellite navigation information provided by external time system frequency standard equipment;

It has the function of ground self-alignment and supports the function of air alignment; It has functions such as power-on self-test, periodic self-test, status report, installation error compensation, and non-volatile storage.

System composition and dimensions

The whole system is composed of two parts: the inertial navigation main instrument and the inertial navigation main instrument bracket. Among them, the external dimensions of the main instrument are as follows:

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Main technical indicators of the system

System accuracy System Accuracy	Pure Inertial Navigation/Pure Inertial Navigation	0.8 nmile/1h, CEP
	Integrated Navigation/Navigation with GNSS	≤5m , 1σ _
	Heading angle /Heading	0.03°, RMS
	Horizontal attitude (roll and pitch) Horizontal Attitude (roll & pitch)	0.006°, RMS

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	Pure Inertial Velo	1 .5 m/s , RMS	
	GNSS Integrated navigation Velocity		0. 1 m/s , RMS
Indicators of inertial devices Gyro and Accelerometer Parameters		Range/Range	± 6 00 deg/s
	laser gyroscope – Gyroscope	Bias Stability	≤0.005 deg/h, 1σ
		Zero bias repeatability Bias Repeatability	≤0.005 deg/h, 1σ
		Scale Factor non-linearity	5 ppm
	Accelerometer Accelerometer	Range/Range	± 15g _
		Bias Stability	≤10μg , 1σ
		Zero bias repeatability Bias Repeatability	≤10μg , 1σ
		Scale Factor non-linearity	15 ppm
	Cold Start		≤ 8 min
align time Alignment Time	Re-Start		≤ 5min _
	Air/In-Flight Start		≤10min
	Supply voltage/Voltage		more than 10h
	Power Consumption		≤ 40W @ 24VDC
Interface Features interface	Electrical interface/Electrical		RS232 × 2 RS422 × 3 CAN × 2 Ethernet × 1 1pps × 1
	Data Update Rate (configurable)		200Hz@115.2kbps _
	Operating Tempe	erature	-40°C~+65°C
Use environment Environmental	Storage temperature/Storage Temperature		-55°C~+85°C
	Use Altitude/Altitude		20000m
	Humidity		95% (+25°C)
	Vibration/Vibration		5g @ 20~2000Hz
	Shock/Shock		40 g, 11 ms, 1/2 Sine
Physical properties Physical	Dimensions/ Size	140 x 233 x 112 mm	



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Weight/ Weight	3 kg

Note: The structure can be customized according to the user's requirements.